Chapter 1:

F = ma $1 J = 1 N \cdot m$ $\rho = \frac{m}{V}$ (kg/m³) Density: $v = \frac{V}{m} = \frac{1}{\rho}$ $SG = \frac{\rho}{\rho_{H_2O}}$ Specific gravity: $\gamma_s = \rho g$ (N/m³) Specific weight: T = a + bP $T(K) = T(^{\circ}C) + 273.15$ $T(R) = T(^{\circ}F) + 459.67$ T(R) = 1.8T(K) $T(^{\circ}F) = 1.8T(^{\circ}C) + 32$ $\Delta T(\mathbf{K}) = \Delta T(^{\circ}\mathbf{C})$ $\Delta T(\mathbf{R}) = \Delta T(^{\circ}\mathbf{F})$ $P_{\text{gage}} = P_{\text{abs}} - P_{\text{atm}}$ $P_{\rm vac} = P_{\rm atm} - P_{\rm abs}$ $\Delta P = P_2 - P_1 = -\rho g \, \Delta z = -\gamma_s \, \Delta z$ $P_{\text{below}} = P_{\text{above}} + \rho g |\Delta z| = P_{\text{above}} + \gamma_s |\Delta z|$ $P = P_{\text{atm}} + \rho g h$ or $P_{\text{gage}} = \rho g h$ $\frac{dP}{dz} = -\rho g$

$$\Delta P = P_2 - P_1 = -\int_1^2 \rho g \, dz$$

$$P_1 = P_2 \quad \rightarrow \quad \frac{F_1}{A_1} = \frac{F_2}{A_2} \quad \rightarrow \quad \frac{F_2}{F_1} = \frac{A_2}{A_1}$$

$$P_{\text{atm}} = \rho g h \qquad P_2 = P_{\text{atm}} + \rho g h$$

$$P_1 + \rho_1 g(a + h) - \rho_2 g h - \rho_1 g a = P_2$$

$$P_1 - P_2 = (\rho_2 - \rho_1) g h$$

Chapter 2:

 $e = \frac{E}{m} \quad (kJ/kg)$ $KE = m\frac{V^2}{2} \quad (kJ) \qquad ke = \frac{V^2}{2} \quad (kJ/kg)$ $PE = mgz \quad (kJ) \qquad pe = gz \quad (kJ/kg)$ $E = U + KE + PE = U + m\frac{V^2}{2} + mgz \quad (kJ)$ $e = u + ke + pe = u + \frac{V^2}{2} + gz \quad (kJ/kg)$ $Mass flow rate: \qquad \dot{m} = \rho \dot{V} = \rho A_c V_{avg} \quad (kg/s)$ $Energy flow rate: \qquad \dot{E} = \dot{m}e \quad (kJ/s \text{ or } kW)$ $e_{mech} = \frac{P}{\rho} + \frac{V^2}{2} + gz$ $\dot{E}_{mech} = \dot{m}e_{mech} = \dot{m} \left(\frac{P}{\rho} + \frac{V^2}{2} + gz\right)$

$$\begin{split} \Delta e_{\text{mech}} &= \frac{P_2 - P_1}{\rho} + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1) \quad (\text{kJ/kg}) \\ \Delta \dot{E}_{\text{mech}} &= \dot{m} \Delta e_{\text{mech}} = \dot{m} \left(\frac{P_2 - P_1}{\rho} + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1) \right) \quad (\text{kW}) \\ q &= \frac{Q}{m} \quad (\text{kJ/kg}) \\ Q &= \int_{t_1}^{t_2} \dot{Q} \, dt \quad (\text{kJ}) \\ Q &= \dot{Q} \Delta t \quad (\text{kJ}) \\ w &= \frac{W}{m} \quad (\text{kJ/kg}) \\ \int_{1}^{2} dV &= V_2 - V_1 = \Delta V \\ \int_{1}^{2} \delta W &= W_{12} \quad (not \Delta W) \\ \dot{W}_e &= \mathbf{V}I \quad (\text{W}) \\ W_e &= \int_{1}^{2} \mathbf{V}I \, dt \quad (\text{kJ}) \\ W &= Fs \quad (\text{kJ}) \\ W &= Fs \quad (\text{kJ}) \\ W &= Fs \quad (\text{kJ}) \\ W &= fr \rightarrow F = \frac{T}{r} \end{split}$$

$$s = (2\pi r)n$$

$$W_{sh} = Fs = \left(\frac{T}{r}\right)(2\pi m) = 2\pi nT \quad (kJ)$$

$$\dot{W}_{sh} = 2\pi nT \quad (kW)$$

$$\delta W_{spring} = F dx$$

$$F = kx \quad (kN)$$

$$W_{spring} = \frac{1}{2}k(x_{2}^{2} - x_{1}^{2}) \quad (kJ)$$

$$W_{elastic} = \int_{1}^{2} F dx = \int_{1}^{2} \sigma_{n} A dx \quad (kJ)$$

$$W_{surface} = \int_{1}^{2} \sigma_{s} dA \quad (kJ)$$

$$\Delta E_{system} = E_{final} - E_{initial} = E_{2} - E_{1}$$

$$\Delta E = \Delta U + \Delta KE + \Delta PE$$

$$\Delta U = m(u_{2} - u_{1})$$

$$\Delta KE = \frac{1}{2}m(V_{2}^{2} - V_{1}^{2})$$

$$\Delta PE = mg(z_{2} - z_{1})$$

$$E_{in} - E_{out} = (Q_{in} - Q_{out}) + (W_{in} - W_{out}) + (E_{mass,in} - E_{mass,out}) = \Delta E_{system}$$

$$E_{in} - E_{out} = \Delta E_{system} (kJ)$$
Rate of net energy transfer
$$kinetic, potential, etc., energies$$

$$Q = \dot{Q} \Delta t, \quad W = \dot{W} \Delta t, \text{ and } \Delta E = (dE/dt) \Delta t \quad (kJ)$$

$$e_{in} - e_{out} = \Delta e_{system} \quad (kJ/kg)$$

$$\delta E_{in} - \delta E_{out} = dE_{system} \quad \text{or} \quad \delta e_{in} - \delta e_{out} = de_{system}$$

$$W_{net,out} = Q_{net,in} \quad \text{or} \quad \dot{W}_{net,out} = \dot{Q}_{net,in} \quad (\text{for a cycle})$$
Efficiency = $\frac{\text{Desired output}}{\text{Required input}}$

$$\eta_{combustion} = \frac{Q}{\text{HV}} = \frac{\text{Amount of heat released during combustion}}{\text{Heating value of the fuel burned}}$$

$$\eta_{overall} = \eta_{combustion} \eta_{thermal} \eta_{generator} = \frac{\dot{W}_{net,electric}}{\text{HHV} \times \dot{m}_{fuel}}$$

$$\eta_{mech} = \frac{\text{Mechanical energy output}}{\text{Mechanical energy input}} = \frac{E_{mech,in}}{E_{mech,in}} = 1 - \frac{E_{mech,ios}}{E_{mech,in}}$$

$$\eta_{pump} = \frac{\text{Mechanical energy increase of the fluid}}{\text{Mechanical energy input}} = \frac{\dot{W}_{shaft,out}}{|\Delta \dot{E}_{mech,fluid}|} = \frac{\dot{W}_{urbine}}{\dot{W}_{urbine,e}}$$

$$Motor: \qquad \eta_{motor} = \frac{\text{Mechanical energy output}}{\text{Electric power input}} = \frac{\dot{W}_{shaft,out}}{\dot{W}_{elect,in}}$$

$$\eta_{pump-motor} = \eta_{pump}\eta_{motor} = \frac{\dot{W}_{pump,u}}{\dot{W}_{elect,in}} = \frac{\dot{W}_{elect,out}}{\dot{W}_{shaft,in}}$$

$$\eta_{turbine-gen} = \eta_{turbine}\eta_{generator} = \frac{\dot{W}_{pump,u}}{\dot{W}_{elect,in}} = \frac{\dot{W}_{elect,out}}{\dot{W}_{elect,in}}$$

Chapter 3:

h = u + Pv (kJ/kg) $H = U + PV \quad (kJ)$ $x = \frac{m_{\text{vapor}}}{m_{\text{total}}}$ $v_{\text{avg}} = v_f + x v_{fg} \quad (\text{m}^3/\text{kg})$ $x = \frac{V_{\text{avg}} - V_f}{V_{f_p}}$ $u_{avg} = u_f + x u_{fg}$ (kJ/kg) $h_{\text{avg}} = h_f + x h_{fg}$ (kJ/kg) $y \cong y_{f@T}$ $h \cong h_{f @ T} + v_{f @ T} (P - P_{sat @ T})$ Pv = RTm = MN (kg) $V = mv \longrightarrow PV = mRT$ $mR = (MN)R = NR_{\mu} \longrightarrow PV = NR_{\mu}T$ $\frac{P_1 V_1}{T_2} = \frac{P_2 V_2}{T_2}$ $V = N\overline{v} \longrightarrow P\overline{v} = R_{\mu}T$

Chapter 4:

$$W_b = \int_1^2 P \, dV$$

(2) Isobaric process

(1) General

$$W_b = P_0(V_2 - V_1)$$
 ($P_1 = P_2 = P_0 = \text{constant}$)

(3) Polytropic process $P \downarrow U = P \downarrow U$

$$W_b = \frac{P_2 V_2 - P_1 V_1}{1 - n} \quad (n \neq 1) \qquad (P V^n = \text{constant})$$

(4) Isothernal process of an ideal gas

$$W_b = P_1 V_1 \ln \frac{V_2}{V_1}$$

= $mRT_0 \ln \frac{V_2}{V_1}$ ($PV = mRT_0 = \text{constant}$)

For a closed system
$$Q - W = \Delta U + \Delta KE + \Delta PE$$

 $W = W_{other} + W_b$

For a constant pressure process $Q - W_{other} = \Delta H + \Delta KE + \Delta PE$

$$\Delta u = u_2 - u_1 = \int_1^2 c_v(T) \, dT \cong c_{v,\text{avg}}(T_2 - T_1)$$
$$\Delta h = h_2 - h_1 = \int_1^2 c_p(T) \, dT \cong c_{p,\text{avg}}(T_2 - T_1)$$

For an ideal gas $c_p = c_v + R$ Specific heat ratio $k = \frac{c_p}{c_v}$

For incompressible substances (liquids and solids) $c_p = c_v = c$

The Δu and Δh of incompressible substances $\Delta u = \int_{1}^{2} c(T) dT \approx c_{avg}(T_2 - T_1)$ $\Delta h = \Delta u + v \Delta P$

 $m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \text{ and } \dot{m}_{\rm in} - \dot{m}_{\rm out} = dm_{\rm system}/dt$ $\dot{m} = \rho VA$ $\dot{\nu} = VA = \dot{m}/\rho$ $\theta = h + \text{ke} + \text{pe} = h + \frac{V^2}{2} + gz$ $\underbrace{E_{\rm in} - E_{\rm out}}_{\text{Rate of net energy transfer}} = \underbrace{dE_{\rm system}/dt}_{\text{Rate of change in internal, kinetic, potential, etc., energies}}$

$$\dot{Q} - \dot{W} = \sum_{\text{out}} \dot{m} \begin{pmatrix} h + \frac{V^2}{2} + gz \end{pmatrix} - \sum_{\text{in}} \dot{m} \begin{pmatrix} h + \frac{V^2}{2} + gz \end{pmatrix}$$
for each exit
$$\int_{\text{for each exit}} h = \frac{1}{2} \int_{\text{for each inlet}} h = \frac{1}{2$$

$$\dot{m}_1 = \dot{m}_2 \longrightarrow \frac{1}{v_1} V_1 A_1 = \frac{1}{v_2} V_2 A_2$$
$$\dot{Q} - \dot{W} = \dot{m} \left[h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1) \right]$$

General Steady State Energy Equation:

$$0 = \frac{\dot{Q}_{cv}}{\dot{m}} - \frac{\dot{W}_{cv}}{\dot{m}} + (h_1 - h_2) + \frac{(V_1^2 - V_2^2)}{2} + g(z_1 - z_2)$$
General Nozzle Eq: $\dot{W}_{cv} = \dot{m}(h_1 - h_2)$
General Turbine Eq: $\frac{\dot{W}_{out}}{\dot{m}}(h_1 - h_2) + \left(\frac{V_1^2 - V_2^2}{2}\right)$
General Compressor and Pump Eq: $\dot{W}_{cv} = \dot{m}(h_1 - h_2)$
General Heat Exchanger Eq.: $0 = \sum_i \dot{m}_i h_i - \sum_e \dot{m}_e h_e$
General Throttling and Valves Eq.: $h_1 = h_2$

$$\eta_{\rm th} = \frac{W_{\rm net,out}}{Q_H} \quad \text{or} \quad \eta_{\rm th} = 1 - \frac{Q_L}{Q_H}$$

$$\operatorname{COP}_{\rm R} = \frac{Q_L}{Q_H - Q_L} = \frac{1}{Q_H/Q_L - 1}$$

$$\operatorname{COP}_{\rm HP} = \frac{Q_H}{Q_H - Q_L} = \frac{1}{1 - Q_L/Q_H}$$

$$\eta_{\rm th,rev} = 1 - \frac{T_L}{T_H} \qquad \operatorname{COP}_{\rm R,rev} = \frac{1}{T_H/T_L - 1} \qquad \operatorname{COP}_{\rm HP,rev} = \frac{1}{1 - T_L/T_H}$$

$$\left(\frac{Q_H}{Q_L}\right)_{\rm rev} = \frac{T_H}{T_L}$$

 $dS = \left(\frac{\delta Q}{T}\right)_{\text{int rev}}$ (kJ/K) $\Delta S = \frac{Q}{T_0}$ (kJ/K) $\Delta S_{\rm sys} = S_2 - S_1 = \int_{-\infty}^{2} \frac{\delta Q}{T} + S_{\rm gen}$ $\Delta S_{\text{isolated}} \ge 0$ $S_{\text{gen}} = \Delta S_{\text{total}} = \Delta S_{\text{sys}} + \Delta S_{\text{surr}} \ge 0$ $\Delta S = m\Delta s = m(s_2 - s_1) \quad \text{(kJ/K)}$ *Isentropic process:* $\Delta s = 0$ or $s_2 = s_1$ (kJ/kg·K) $Q_{\rm int\,rev} = \int_{-\infty}^{\infty} T dS$ (kJ) $Q_{\text{int rev}} = T_0 \Delta S$ (kJ) $q_{\text{int rev}} = T_0 \Delta s$ (kJ/kg) Tds = du + Pdv (kJ/kg) $\begin{array}{cccc} h = u + Pv & \longrightarrow & dh = du + Pdv + vdP \\ (\text{Eq. 7-23}) & \longrightarrow & Tds = du + Pdv \end{array} \right\} Tds = dh - vdP$ $ds = \frac{du}{T} + \frac{P dv}{T}$ $ds = \frac{dh}{T} - \frac{v dP}{T}$ Liquids, solids: $s_2 - s_1 = \int_{-\infty}^{2} c(T) \frac{dT}{T} \approx c_{\text{avg}} \ln \frac{T_2}{T_2}$ (kJ/kg·K) Isentropic: $s_2 - s_1 = c_{avg} \ln \frac{T_2}{T_2} = 0 \longrightarrow T_2 = T_1$

For an ideal gas

$$ds = c_{v}\frac{dT}{T} + R\frac{dv}{v} \qquad s_{2} - s_{1} = \int_{1}^{2} c_{v}(T)\frac{dT}{T} + R\ln\frac{v_{2}}{v_{1}} \qquad s_{2} - s_{1} = \int_{1}^{2} c_{p}(T)\frac{dT}{T} - R\ln\frac{P_{2}}{P_{1}}$$

For an ideal gas assuming constant heat capacities

$$s_{2} - s_{1} = c_{v,avg} \ln \frac{T_{2}}{T_{1}} + R \ln \frac{v_{2}}{v_{1}} \quad (kJ/kg\cdot K)$$

$$s_{2} - s_{1} = c_{p,avg} \ln \frac{T_{2}}{T_{1}} - R \ln \frac{P_{2}}{P_{1}} \qquad (kJ/kg\cdot K)$$

For an ideal gas with variable heat capacities

$$s_2 - s_1 = s_2^\circ - s_1^\circ - R \ln \frac{P_2}{P_1}$$
 (kJ/kg·K)

Isentropic process of ideal gases

$$\begin{pmatrix} \frac{T_2}{T_1} \end{pmatrix}_{s=\text{const.}} = \begin{pmatrix} \frac{v_1}{v_2} \end{pmatrix}^{k-1} \quad \text{(ideal gas)}$$

$$\begin{pmatrix} \frac{T_2}{T_1} \end{pmatrix}_{s=\text{const.}} = \begin{pmatrix} \frac{P_2}{P_1} \end{pmatrix}^{(k-1)/k} \quad \text{(ideal gas)}$$

$$\begin{pmatrix} \frac{P_2}{P_1} \end{pmatrix}_{s=\text{const.}} = \begin{pmatrix} \frac{v_1}{v_2} \end{pmatrix}^k \quad \text{(ideal gas)}$$

$$s_2^{\circ} = s_1^{\circ} + R \ln \frac{P_2}{P_1}$$

$$\begin{pmatrix} \frac{P_2}{P_1} \end{pmatrix}_{s=\text{const.}} = \frac{P_{r2}}{P_{r1}} \quad \begin{pmatrix} \frac{v_2}{v_1} \end{pmatrix}_{s=\text{const.}} = \frac{v_{r2}}{v_{r1}}$$

Reversible steady flow work

$$w_{\text{rev}} = -\int_{1}^{2} v dP - \Delta ke - \Delta pe \quad (kJ/kg)$$
$$w_{\text{rev,in}} = \int_{1}^{2} v dP + \Delta ke + \Delta pe$$

 $w_{\text{rev}} = -v(P_2 - P_1) - \Delta \text{ke} - \Delta \text{pe}$ (kJ/kg)

Compressor work

Isentropic
$$w_{\text{comp,in}} = \frac{kR(T_2 - T_1)}{k - 1} = \frac{kRT_1}{k - 1} \left[\left(\frac{P_2}{P_1} \right)^{(k-1)/k} - 1 \right]$$

Polytropic $w_{\text{comp,in}} = \frac{nR(T_2 - T_1)}{n - 1} = \frac{nRT_1}{n - 1} \left[\left(\frac{P_2}{P_1} \right)^{(n-1)/n} - 1 \right]$
Isothermal $w_{\text{comp,in}} = RT \ln \frac{P_2}{P_1}$

Isothermal

$$P_x = (P_1 P_2)^{1/2}$$
 or $\frac{P_x}{P_1} = \frac{P_2}{P_x}$

Isentropic efficiency of turbines

$$\eta_T = \frac{\text{Actual turbine work}}{\text{Isentropic turbine work}} = \frac{w_a}{w_s} \qquad \eta_T \cong \frac{h_1 - h_{2a}}{h_1 - h_{2s}}$$

Isentropic efficiency of compressors and pumps

$$\eta_C = \frac{\text{Isentropic compressor work}}{\text{Actual compressor work}} = \frac{w_s}{w_a} \qquad \eta_C \cong \frac{h_{2s} - h_1}{h_{2a} - h_1}$$

Isentropic efficiency of nozzle

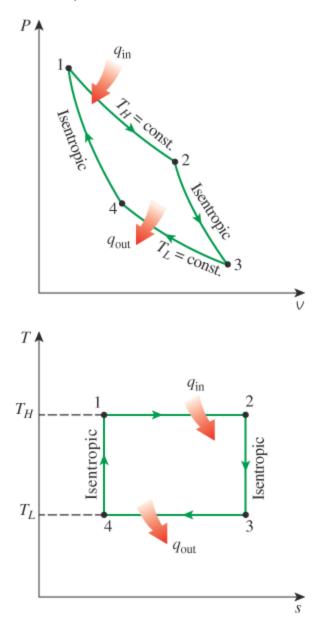
$$\eta_N = \frac{\text{Actual KE at nozzle exit}}{\text{Isentropic KE at nozzle exit}} = \frac{V_{2a}^2}{V_{2s}^2} \qquad \eta_N \cong \frac{h_1 - h_{2a}}{h_1 - h_{2s}}$$
$$\Delta S_{\text{system}} = S_{\text{final}} - S_{\text{initial}} = S_2 - S_1$$

Entropy transfer by heat transfer:
$$S_{heat} = \frac{Q}{T}$$
 (T = constant)Entropy transfer by work: $S_{work} = 0$ Entropy transfer by mass flow: $S_{mass} = ms$

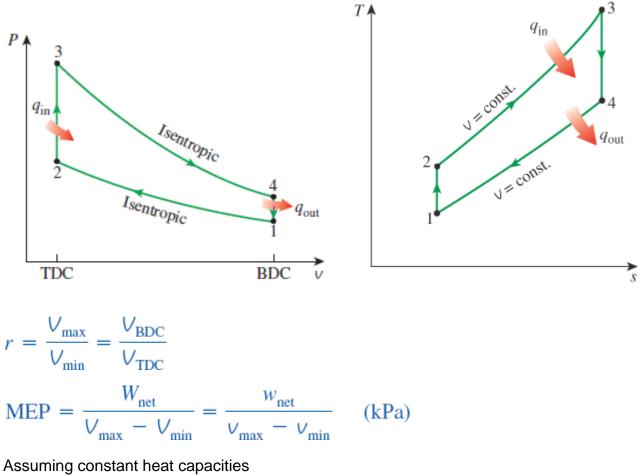
<u>Chapter 9</u>

$$\eta_{\text{th,Carnot}} = 1 - \frac{T_L}{T_H}$$
 $\eta_{\text{th}} = \frac{W_{\text{net}}}{Q_{\text{in}}}$ or $\eta_{\text{th}} = \frac{w_{\text{net}}}{q_{\text{in}}}$

Carnot cycle

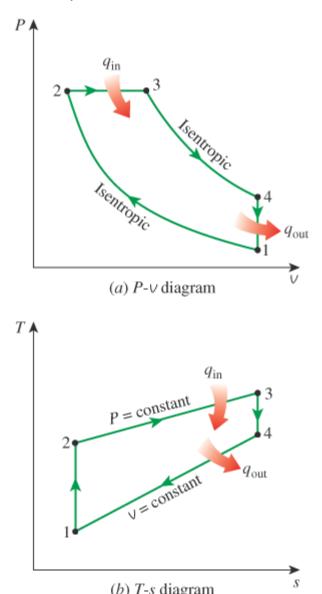






$$\eta_{\rm th,Otto} = 1 - \frac{1}{r^{k-1}}$$

Diesel cycle

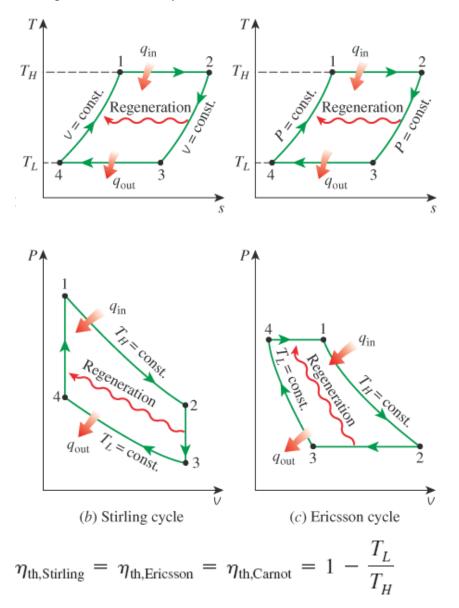


(b) T-s diagram

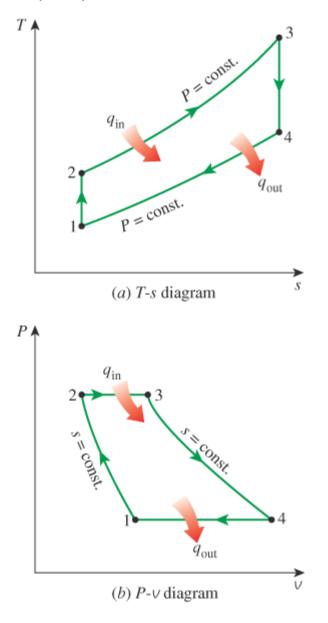
Assuming constant heat capacities

$$\eta_{\text{th,Diesel}} = 1 - \frac{1}{r^{k-1}} \left[\frac{r_c^k - 1}{k(r_c - 1)} \right] \qquad r_c = \frac{V_3}{V_2} = \frac{V_3}{V_2}$$

Stirling and Ericsson cycles



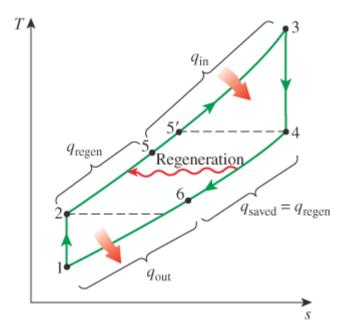
Brayton cycle



Assuming constant heat capacities

$$\eta_{\text{th,Brayton}} = 1 - \frac{1}{r_p^{(k-1)/k}}$$
$$r_p = \frac{P_2}{P_1}$$

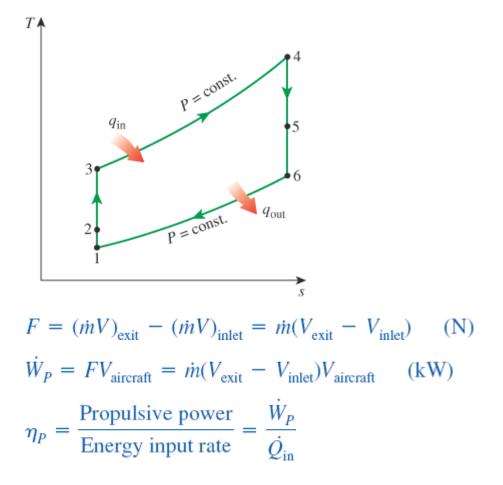
Brayton cycle with regeneration



$$\epsilon = \frac{q_{\text{regen,act}}}{q_{\text{regen,max}}} = \frac{h_5 - h_2}{h_4 - h_2}$$

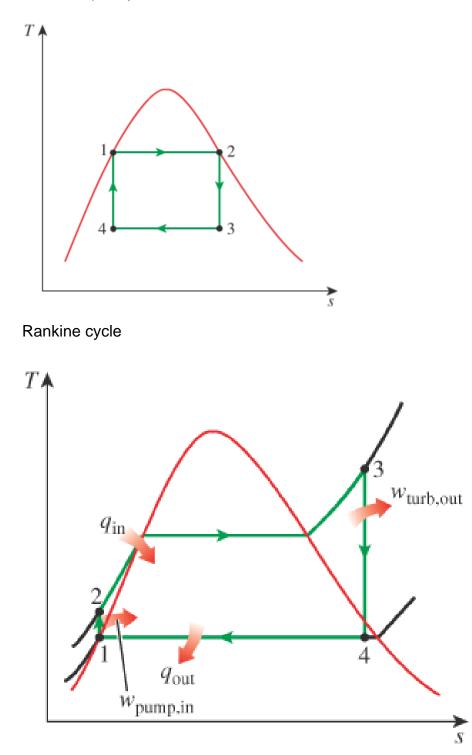
$$\eta_{\text{th,regen}} = 1 - \left(\frac{T_1}{T_3}\right) (r_p)^{(k-1)/k}$$

Ideal jet-propulsion cycle



<u>Chapter 10</u>

Carnot vapor cycle



$$Pump (q = 0): \qquad \qquad w_{\text{pump,in}} = h_2 - h_1$$

or,

$$w_{\text{pump,in}} = v(P_2 - P_1)$$

where

 $h_1 = h_{f @ P_1}$ and $v \cong v_1 = v_{f @ P_1}$ $q_{in} = h_3 - h_2$

Turbine
$$(q = 0)$$
: $w_{turb,out} = h_3 - h_4$

Condenser (w = 0):

Boiler (w = 0):

$$q_{\rm out} = h_4 - h_1$$

$$\eta_{\text{th}} = \frac{w_{\text{net}}}{q_{\text{in}}} = 1 - \frac{q_{\text{out}}}{q_{\text{in}}}$$

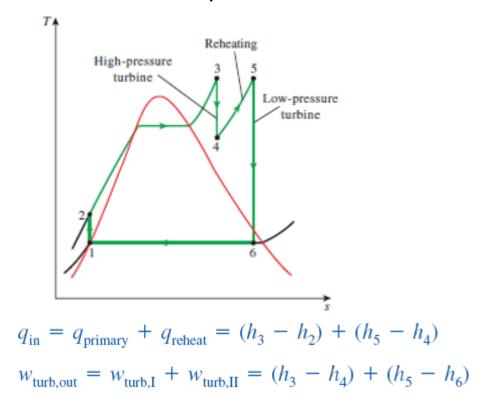
$$w_{\text{net}} = q_{\text{in}} - q_{\text{out}} = w_{\text{turb,out}} - w_{\text{pump,in}}$$

$$\eta_{\text{th}} = \frac{3412 \text{ (Btu/kWh)}}{\text{Heat rate (Btu/kWh)}}$$

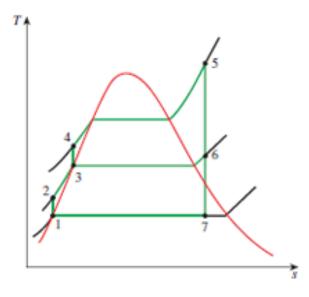
$$\eta_P = \frac{w_s}{w_a} = \frac{h_{2s} - h_1}{h_{2a} - h_1}$$

$$\eta_T = \frac{w_a}{w_s} = \frac{h_3 - h_{4a}}{h_3 - h_{4s}}$$

The ideal reheat Rankine cycle



The ideal regenerative Rankine cycle with an open feedwater heater



$$q_{in} = h_5 - h_4$$

$$q_{out} = (1 - y)(h_7 - h_1)$$

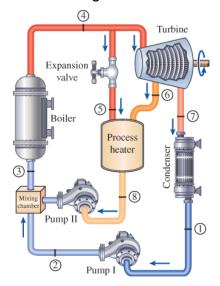
$$w_{turb,out} = (h_5 - h_6) + (1 - y)(h_6 - h_7)$$

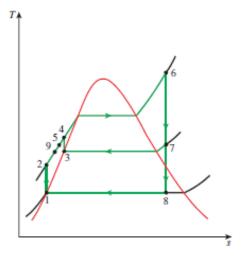
 $w_{\text{pump,in}} = (1 - y)w_{\text{pump I,in}} + w_{\text{pump II,in}}$

$$y = \dot{m}_6 / \dot{m}_5 \quad \text{(fraction of steam extracted)}$$
$$w_{\text{pump I,in}} = v_1 (P_2 - P_1)$$
$$w_{\text{pump II,in}} = v_3 (P_4 - P_3)$$

$$\dot{E}_{in} = \dot{E}_{out} \longrightarrow \sum_{in} \dot{m}h = \sum_{out} \dot{m}h$$
$$yh_6 + (1 - y)h_2 = 1(h_3)$$
$$y = \frac{h_3 - h_2}{h_6 - h_2}$$

The ideal regenerative Rankine cycle with a closed feedwater heater





Cogeneration

$$Q_{in} = \dot{m}_3(h_4 - h_3)$$

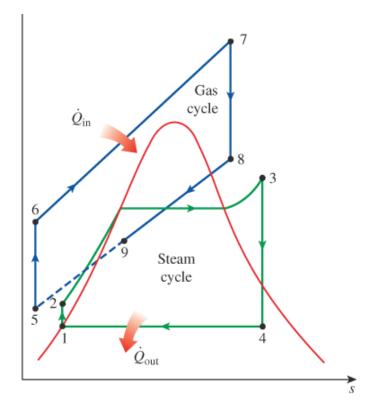
$$\dot{Q}_{out} = \dot{m}_7(h_7 - h_1)$$

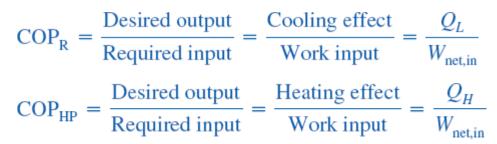
$$\dot{Q}_p = \dot{m}_5 h_5 + \dot{m}_6 h_6 - \dot{m}_8 h_8$$

$$\dot{W}_{turb} = (\dot{m}_4 - \dot{m}_5)(h_4 - h_6) + \dot{m}_7(h_6 - h_7)$$

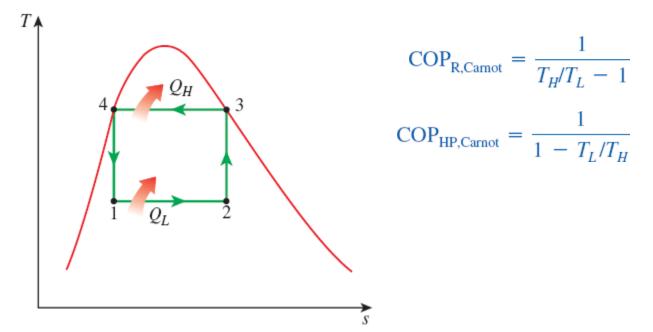
$$\epsilon_u = \frac{\text{Net power output} + \text{Process heat delivered}}{\text{Total heat input}} = \frac{\dot{W}_{\text{net}} + \dot{Q}_p}{\dot{Q}_{\text{in}}}$$

Combined gas-vapor power cycle

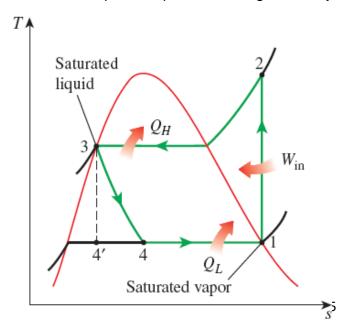




Reversed Carnot cycle



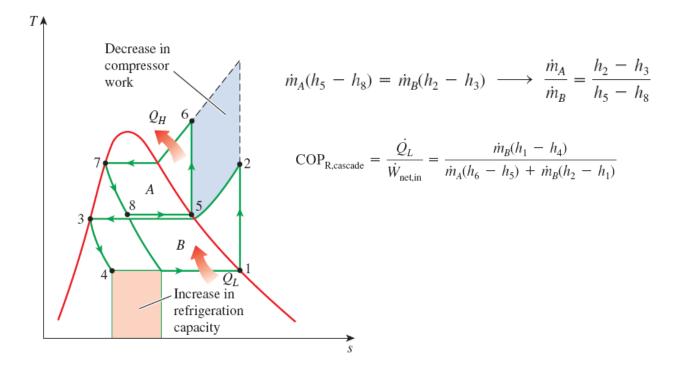
The ideal vapor-compression refrigeration cycle



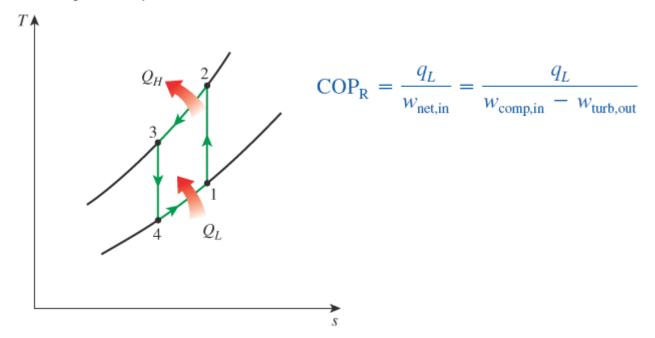
$$\operatorname{COP}_{\mathrm{R}} = \frac{q_L}{w_{\mathrm{net,in}}} = \frac{h_1 - h_4}{h_2 - h_1}$$

$$\text{COP}_{\text{HP}} = \frac{q_H}{w_{\text{net,in}}} = \frac{h_2 - h_3}{h_2 - h_1}$$

Cascade refrigeration cycles



Gas refrigeration cycles



Absorption refrigeration cycles

